

Gas bubble dynamics in airlift photo-bioreactors for microalgae cultivation by level set methods

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Abstract

Coupled level set and CFD (computational fluid dynamics) methods are adopted in this work to track the moving gas-liquid interfaces in the riser of an external loop airlift photobioreactor (ALR) in which microalgae are used to produce biofuels and capture CO₂ from flue-gas. Modeling the behavior of gas bubbles is a crucial aspect for the fine-tuning of the operation of the reactor when inserted into a closed-loop biorefinery at the pilot-scale. The experimental data used for simulation were completely acquired or calculated from hydrodynamic experimental campaigns carried out on the ALRs. The rise, coalescence, and shape dynamics of the bubbles of the flue-gas are simulated in a rectangular domain representing the vertical section of the ALR riser. Different correction approaches, such as the conservative level set method (CLSM), are proposed to face the volume loss characteristic of LSM. Computational results evidenced strong agreement with the experimental data (bubble shapes and trajectories). The physically-based CLSM model was then effectively used for the fine-tuning of the multiphase flow regime inside the ALRs, suggesting operating conditions for the outdoor cultivation with Reynolds number = 10000 – 11000, Sherwood number between 1400 and 1800, and spherical-caps bubbles in the upper half of the riser, mildly churning the microalgae while avoiding damages to their cells.

Keywords: Airlift; CO₂ capture, bubble dynamics; level set method; biorefineries; microalgae

26 **1. Introduction**

27 Biofuels represent a key alternative to reduce the carbon footprint of the energy sector. They are
28 particularly appreciated in the transport sector, as they allow the exploitation of existing distribution
29 and transport infrastructures [1] without prolonging the supply times. Particular interest in recent
30 years has been dedicated to biodiesel produced by microalgae, since these organisms feed on nitrogen
31 and phosphorus, being therefore usable for the bio-fixation of liquid waste streams (wastewater) and,
32 through photosynthesis, they can sequester CO₂ from exhaust gases from the combustion of biofuels
33 themselves, thus being promising for use in closed cycle biorefineries [2]. They are able to achieve
34 high lipid productivity and do not create competitiveness on land use and/or food in the synthesis of
35 biodiesel [3].

36 This type of biofuel production is not yet widely applied for marketing [4] as microalgae provide low
37 yields that are not compatible with the high productivity required [5]; besides, large-scale production
38 requires oil extraction and transformation processes that significantly affect costs [6]. Interesting
39 options are the biorefineries, where first/second generation biofuel production systems are coupled
40 with the cultivation of microalgae within integrated plants [7]. To this end, at the ECPL laboratory of
41 the University of Genoa, a pilot-scale biorefinery has been designed. It uses waste frying oil,
42 microalgae, and lignocellulosic biomass for the combined production of biodiesel and syngas and
43 their transformation into energy, with the aim to minimize waste streams and partially sequestering
44 CO₂ from flue-gas [8]. The plant is divided into three main sub-processes: the first is the synthesis of
45 biodiesel by transesterification, the second is the production of syngas by gasification of biomass
46 added with crude glycerol, and the third, the most critical one, involves the cultivation of *Chlorella*
47 *vulgaris* in airlift photobioreactors (ALRs). Microalgae are fed by the wastewater from the
48 transesterification process rich in glycerol and by the flue-gas produced during the combustion of
49 syngas.

50 Airlift photobioreactors are very flexible and have been investigated for microalgae nurture, with
51 the aim of producing biofuels [9] and/or CO₂ sequestration. In this sense, ALRs, although more
52 complicated to deploy, are flexible and markedly suitable to be incorporated into biorefineries and
53 integrated power generation systems, due to the multitude [10,11] of process parameters that can be
54 used to optimize their operation.

55 The present work deals with the multiscale simulation of the external-loop ALRs employed in the
56 pilot-plant described in [8]. For the growth of microorganisms, airlift reactors are considered to be
57 superior to both tubular bubble reactors and open ponds; it is possible to obtain very fast light/dark
58 cycles and good mixing without high energy demand. However, bubble size is a pivotal aspect as it
59 reflects itself on mass transfer. The optimization of the size and shape of the bubbles ensures a good
60 mass transfer between gas and liquid and guarantees the recirculation of microalgae. Knowledge of
61 the flow regime is thus a piece of crucial information to optimize processes inside ALRs. When
62 biological reactants such as microalgae are used and bubble flow could damage the cells, the
63 possibility to manipulate the flow regime of the bubbles during microalgae growth becomes a major
64 goal. All this motivates the study and simulation of the phenomenon in detail.

65 The scope of this paper is the development of a model based on the Level Set Method (LSM) [12]
66 for the simulation of the bubble dynamics in the ALR riser, as well as its verification and validation
67 with the experimental data collected during some hydrodynamic tests. This modeling approach was
68 chosen since it allows taking into account topology variations, i.e. the possibility of bubble
69 coalescing and splitting. This is a significant aspect, as knowledge of the shape and trajectory of the
70 rising bubbles is crucial to increase CO₂ exchange between exhaust gases and microalgae.

71 Moreover, the representation of the moving interfaces, i.e. the surfaces of the bubbles, is achieved
72 with a continuous implicit function defined on the whole domain. This implies a simplification as
73 the numerical calculation of a solution does not require the development of complicated algorithms
74 for the reconstruction of the front.

75 The multiphase model, after validation, was used for the fine-tuning of the ALRs operating
76 conditions for the outdoor cultivation tests.

77

78 **2. Materials and methods**

79 **2.1. Experimental reactor & measurements**

80 Four external loop airlift reactors (EL-ALRs) were designed and assembled to conduct an
81 experimental campaign principally apt to gauge the main dynamic parameters [8] and to conduct a
82 rigorous scale-up procedure [13] of the cultivation process. A sketch of one of the EL-ALRs,
83 annotated with the main dimensions, is illustrated in Figure 1. The construction material is
84 transparent PolyMethyl MethAcrylate (PMMA) as regards the riser and the downcomer and
85 PolyVinyl Chloride (PVC) for the horizontal collectors. This choice is made since light availability
86 and dark/light cycle alternation play a decisive role in algae cultivation.

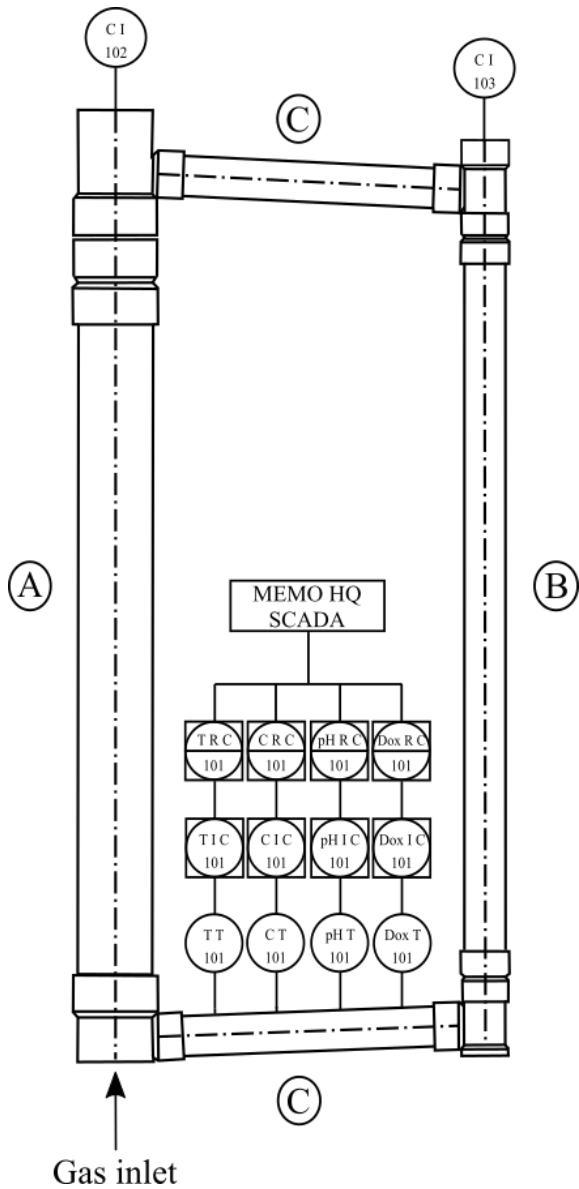
87 Gas is vertically insufflated in the photobioreactor through a sparger made of a perforated metal
88 plate coupled with a porous sponge as a diffuser, located on the base of the riser. The ratio between
89 riser and downcomer diameters is 2.2, falling within the [1.0, 3.0] range recommended in literature
90 as the optimum one for airlift bioreactor design [14]. The volume of each reactor amounts to 10.5 L.
91 For the hydrodynamic, mass transfer, and cultivation experimental campaigns, both compressed air
92 and carbon dioxide were used as the gas phase and pumped inside the reactor. Water and water plus
93 glycerol (coming from the transesterification process of oil) constituted the liquid phase, while the
94 third solid phase, dispersed in water, was constituted by microalgae *Chlorella vulgaris*.

95 Temperature, pH, conductivity, and dissolved oxygen (DO) probes plus microcontrollers (B&C
96 Electronics, Italy) were installed in the midsection of the lower collector; conductivity was also
97 detected at the top of both the riser and of the downcomer (WTW multiparameter portable device
98 3630) during the hydrodynamic campaign.

99 Both gas pressure and consequently mean fluid velocity were varied in the hydrodynamic tests to
100 evaluate optimal residence times. Tracing tests, as suggested in [15], hinging on the injection of
101 NaCl in correspondence with the riser sparger, were executed to compute, through electrical
102 conductivity measures, the mean liquid velocity in the reactor sections. Bubble count and evaluation
103 of bubble shape and dimension were performed by image acquisition (CANON EOS20D camera
104 equipped with a CCD sensor). pH measures were used during these tests as an indirect estimation of
105 carbon dioxide dissolved in circulating water. The data thereby continuously collected were stored
106 in a real-time, remotely accessible database, as the reactor was equipped with Memo HQ SCADA
107 system [16].

108 Generally, temperatures around 15 - 25 °C seems suitable for most algal species, even those which
109 are adapted to grow at colder temperatures [17]. However, since the temperature of the solution
110 inside the photobioreactor can grow in outdoor conditions, during the summer season the
111 implementation of a temperature controller is recommended, even if it was not necessary during the
112 outdoor cultivation tests (carried out between May and June at latitude 44°18' N). Note that the
113 airlift reactor is thought of as an element of a closed-loop cycle, receiving CO₂-rich flue gases as the
114 input gas phase. It follows that there should not be issues of low temperature, so a cooling circuit
115 can be used.

116 *C. vulgaris* can grow in a temperature range of 12 - 26 °C (optimal 18 - 24 °C), and ranges for pH of
117 6.8 - 10.5 (optimal 7.2 - 8.5). It can tolerate high CO₂ concentrations, up to 18% (v/v), with satisfactory
118 fixation and growth rates [18]. Moreover, it can use crude glycerol to increase its lipid content and
119 growth rates [19]. In mixotrophic conditions, glycerol concentration can reach 5 - 10 g L⁻¹.



120

Gas inlet

121

Figure 1. External loop airlift reactor: A – riser; B – downcomer; C – horizontal collectors.

122

In Table 1 the chemical-physical properties of the fluids are reported. The ALR dimensions and the

123

dynamic parameters and operating variables measured during the experimental campaign are

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collected in Table 2. These data are used for simulation.

125

Table 1. Physical-chemical data used in the simulation.

Physical-chemical parameters	Value
Specific gas constant (CO ₂)	188.9 J kg ⁻¹ K ⁻¹
Surface tension coefficient (CO ₂)	72.86 · 10 ⁻³ N m ⁻¹

Liquid density (H ₂ O)	1000 kg m ⁻³
Liquid dynamic viscosity (H ₂ O)	1.002 · 10 ⁻³ Pa s
Gas dynamic viscosity (CO ₂)	1.47 · 10 ⁻⁵ Pa s

126

127 Table 2. Data gathered in the experimental campaigns.

Geometric data	Value
Length of the riser	0.780 m
Diameter of the riser	0.110 m
Length of the downcomer	0.780 m
Diameter of the downcomer	0.050 m
Length of the horizontal collectors	0.385 m
Diameter of the horizontal collectors	0.050 m
Operating variables	Value
Temperature	293 K
Pressure	3 - 4 bar
pH	7.0 - 9
CO ₂	15 - 20 %
NaCl pulse injection	1 - 10 g L ⁻¹
Measured variables	Value
Gas velocity	0.57 - 0.64 m s ⁻¹
Liquid velocity (in the riser)	0.06 - 0.14 m s ⁻¹
Liquid velocity (in the downcomer)	0.17 - 0.39 m s ⁻¹
(min, average, max) diameter of the bubbles	(0.003, 0.007, 0.020) m
Average number of bubbles in the riser	2200 - 2800

128

129

130 **2.2. Mathematical modeling**

131 The key role played by the hydrodynamics of the multi-phase flow for the proper operation of
132 ALRs recommends detailed modeling of this aspect. In fact, it affects the shape and trajectory of the
133 bubbles in the riser, which in turn significantly influences the mass transfer between gas and
134 microalgae and hence the growth of the latter.

135 To build an appropriate model, two essential phenomena have to be taken into account: viscous
136 fluid flow and moving interfaces, in addition to their mutual influence. Fluid flow is described by
137 the non-dimensional form of the incompressible Navier-Stokes equations whereas interfaces are
138 described through an implicit representation by resorting to the level set approach. The link between
139 the two-equation systems is offered by the formulation of physical properties of the considered
140 fluids, i.e. density ρ , dynamic viscosity μ and surface tension, all described as functions of the level
141 set function ϕ .

142 The local formulation of the Navier-Stokes equations (NSE), obtained from momentum balance by
143 resorting to the divergence theorem in conjunction with Reynolds transport theorem, is given by
144 [20,21]

$$\rho \frac{D\mathbf{u}}{Dt} = -\nabla p + \left(\zeta + \frac{1}{3}\mu \right) \nabla(\nabla \cdot \mathbf{u}) + \mu \Delta \mathbf{u} + \boldsymbol{\xi} \quad (1)$$

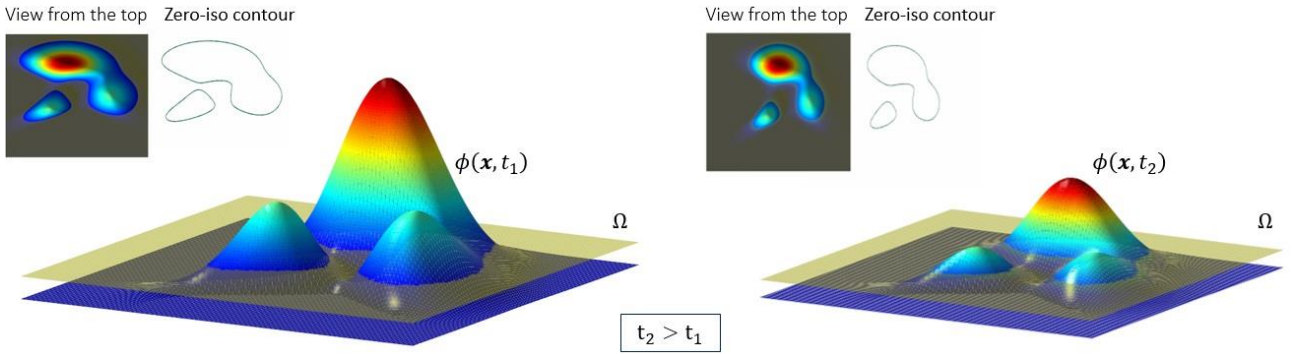
145 where the ratio $D\mathbf{u}/Dt$ denotes the material derivative of the velocity \mathbf{u} . Moreover, ρ is the density,
146 p the pressure of the fluid, μ the dynamic viscosity while ζ represents the volume viscosity (also
147 known as second viscosity coefficient) and $\boldsymbol{\xi}$ other body forces (per unit volume) such as
148 gravitational force per unit volume $\rho\mathbf{g}$.

149 As it is explained in the Supplementary Materials section S1, Eq. (1) can be non-dimensionalized
150 as:

$$\frac{D\mathbf{u}^*}{Dt^*} = -\nabla^* p^* + \frac{1}{Fr^2} \hat{\mathbf{e}}_g + \frac{1}{Re} \Delta^* \mathbf{u}^* \quad (2)$$

151 where all the star superscripts indicate that the variables are non-dimensional, $\hat{\mathbf{e}}_g = \mathbf{g}^*$, Fr is the
 152 Froude number, and Re is the Reynolds number.

153 The Level Set Method (LSM) is based upon a rather simple idea: the interface can be implicitly
 154 thought of as the zero iso-contour of a signed distance field $\phi(\mathbf{x}, t)$ called level set function (see
 155 Figure 2).



156
 157 Figure 2. Illustration of the LSM.

158 Having called $\Omega \subset \mathbb{R}^n$ the space domain and T the final time, the level set function $\phi: \Omega \times [0, T] \rightarrow$
 159 \mathbb{R} is the solution of the Hamilton-Jacobi equation

$$\begin{cases} \frac{\partial \phi(\mathbf{x}, t)}{\partial t} = \mathbf{u}(\mathbf{x}, t) \|\nabla \phi(\mathbf{x}, t)\| \\ \phi(\mathbf{x}, 0) = \phi_0(\mathbf{x}) \end{cases} \quad (3)$$

160 where $\|\cdot\|$ is the Euclidean norm, $\phi_0: \Omega \rightarrow \mathbb{R}$ a given initial condition and $\mathbf{u}: \Omega \times [0, T] \rightarrow \mathbb{R}$ is the
 161 speed vector field. The zero level of ϕ_0 coincides with the initial location of the front and can
 162 therefore be set to the signed distance to it, meaning that the domain Ω is subdivided into two
 163 distinct regions Ω^+ and Ω^- ($\Omega^+ \cap \Omega^- = \emptyset$) by the boundary $\partial\Omega$ between Ω^+ and Ω^- ($\partial\Omega \notin$
 164 $\{\Omega^+, \Omega^-\}$), i.e. the zero level of ϕ_0 .

$$\begin{cases} \phi(\mathbf{x}, t) > 0 & \mathbf{x} \in \Omega^+ \\ \phi(\mathbf{x}, t) = 0 & \mathbf{x} \in \partial\Omega \\ \phi(\mathbf{x}, t) < 0 & \mathbf{x} \in \Omega^- \end{cases} \quad (4)$$

165 The level set approach presupposes the computation of the latter at each time step, following the
 166 resolution of Eq. (3), to obtain the front propagation. However, an anticipated consequence of the
 167 convection term in Eq. (3) is the need for re-initialization of the distance profile, as a loss in the
 168 smoothness of field ϕ is to be expected. More details on the LSM can be found in section S2.

169 The link between the equations for the velocity field (NSE) and the interface tracking (realized
 170 thanks to an implicit representation obtained through the LSM) is offered by the formulation of the
 171 physical properties of the considered fluids (e.g. density and viscosity) as functions of the level set
 172 function ϕ . In this way, all the three essential elements (i.e. the velocity field, a description of the
 173 moving fronts and the coupling between interface tracking, and the velocity field) are effectively
 174 taken into account.

175 When multiphase flows are considered, jump discontinuities are commonly encountered across the
 176 fronts due to the diversity in those physical properties. The use of the smoothed Heaviside function
 177 $H_\varepsilon(\phi)$ allows the definitions of the density ρ and the kinematic viscosity ν of the two-phase flow
 178 system of incompressible fluids as:

$$\rho = \rho_g + (\rho_l - \rho_g)H_\varepsilon(\phi) \quad (5)$$

$$\nu = \nu_g + (\nu_l - \nu_g)H_\varepsilon(\phi) \quad (6)$$

179 being the subscripts g and l tags referring to the gas and liquid phases respectively. Another
 180 quintessential ingredient to model multiphase flows is the surface tension force (per unit length)
 181 ξ_{ST} . In the context of front-tracking methods, the most widespread way to take into account such
 182 parameter is as a force concentrated on the interface [22], i.e.

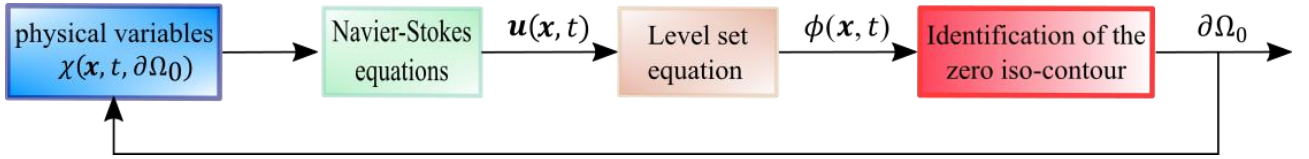
$$\xi_{ST}(\phi) = \sigma\kappa\hat{\mathbf{n}}\delta(\phi) \quad (7)$$

183 where σ is the coefficient of surface tension, $\hat{\mathbf{n}}$ is the normal vector of the moving interfaces, κ is
 184 the curvature, and, for computational purposes, one can use the mollified delta function $\delta_\varepsilon(\phi)$
 185 instead of $\delta(\phi)$. This leads to a modified version of Eq. (2), as ξ needs to be updated as $\xi = \rho\mathbf{g} +$
 186 $\sigma\kappa\hat{\mathbf{n}}\delta(\phi)$.

$$\frac{\partial \mathbf{u}^*}{\partial t^*} + \mathbf{u}^* \cdot \nabla^* \mathbf{u}^* = -\nabla^* p^* + \frac{1}{Fr^2} \hat{\mathbf{e}}_g + \frac{\sigma\kappa\hat{\mathbf{n}}\delta_\varepsilon}{\rho} \frac{L}{U^2} + \frac{1}{Re} \Delta^* \mathbf{u}^* \quad (8)$$

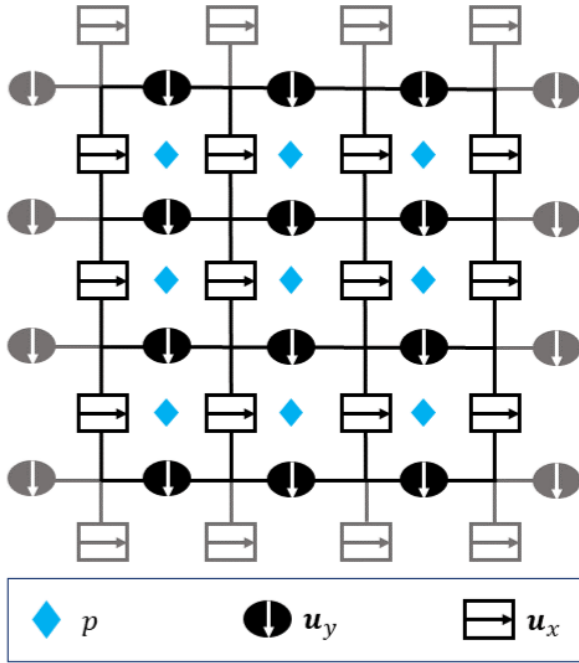
187 2.3. Model implementation

188 The simulation framework was articulated in MATLAB environment, with a cascade model (Figure
 189 3): a first resolution of the flow field modeled with the Navier-Stokes PDEs system and a second
 190 PDE, the Hamilton-Jacobi equation (Eq. (3)), handled to model the moving fronts related to the
 191 process described by the first PDEs system.



192
 193 Figure 3. Numerical model schematization.

194 In particular, bidimensional NSEs in the non-dimensional form are solved by employing Chorin's
 195 projection on a Marker-and-Cell (MAC) staggered grid (Figure 4). The relative code is based on the
 196 NS solver programmed by Seibold [23] and utilizes a three-step semi-implicit scheme for time
 197 discretization (explicit treatment of the nonlinear convective term, implicit handling of the diffusive
 198 term, and pressure correction).



199

200 Figure 4. Staggered grid with boundary cells.

201 The portion concerning the resolution of the Hamilton-Jacobi equation for computing the zero level
 202 set was instead performed by using Mitchell's LSM toolbox [24]. The spatial discretization is
 203 performed by the means of an upwind second-order essentially non-oscillatory (ENO) scheme
 204 whilst time discretization is carried out with a 3-step, second-order total variation diminishing
 205 (TVD) Runge-Kutta scheme. The choice of upwind approximations is motivated by the hyperbolic
 206 nature of the PDE to avoid numerical instabilities. All PDEs were treated by relying on finite
 207 difference approximations.

208 The system equations were numerically solved in the spatial domain $\Omega = \left[-\frac{d_r}{2}, \frac{d_r}{2}\right] \times \left[-\frac{L_r}{2}, \frac{L_r}{2}\right]$,
 209 where L_r is the length of the ALR riser, equal to 0.78 m, and d_r is its diameter, equal to 0.11 m. Ω
 210 was discretized to form a mesh grid of rectangular cells, made up of 300 nodes in each dimension.
 211 Another possibility, whose results are reported in this work, consists of focusing on the bottom
 212 section of the airlift reactor, with the diffuser diameter (equal to the riser diameter) as the
 213 characteristic length. Since the collector diameter d_c is about half the riser diameter ($d_c = 0.05$ m),

214 in this case, a square domain $\Omega = d_r \times d_r$ could better capture both bubble shape modifications and
215 initial behavior in the inlet section, where the diffuser is highly influenced by the lateral collector.
216 Referring for now to the first instance, i.e. Ω as a rectangular domain, the boundary conditions
217 employed to evaluate the velocity field were of the Dirichlet type. No-slip conditions were imposed
218 on three sides of the domain, both in the x and y direction. The right-side wall, on the contrary, was
219 modeled to consider the inflow and outflow determined by the presence of the horizontal collectors.
220 Hence, in correspondence with the openings, a noise-imposed, fairly flat profile velocity
221 distribution typical of the turbulent fluid flow was adopted, while no-slip conditions were used
222 elsewhere.

223

224 **3. Results & discussion**

225 **3.1 Bubble shape and hydrodynamic regime**

226 In order to maximize mass transfer, the maximum surficial area of exchange is required. It is also
227 self-evident that, in a chosen surficial area, the longer the transport phenomenon lasts, the greater is
228 the gain. This means that, in addition to needing suitable forms of the bubbles, it is also necessary to
229 optimize their residence time in the riser, ensuring that appropriate trajectories are achieved.
230 Hydrodynamics of multiphase flow thus holds a controlling influence on mass transport
231 phenomena.

232 As long as the gas inlet velocity is maintained below a threshold value, dependent on the tube
233 geometry, bubbles rise almost individually without significant interactions between them and with
234 narrow bubble size distribution. In this flow condition, known as bubble or homogeneous flow,
235 values of the diameter of the bubble d_b (taken equal to the diameter of a sphere having the same
236 volume as the bubble) generally fall within the range of 1 - 6 mm. The ascent path is mostly
237 rectilinear, with minor transverse and axial oscillations [25]. Whenever the gas phase velocity
238 exceeds the aforementioned threshold, the density of the gaseous fraction in the liquid gradually

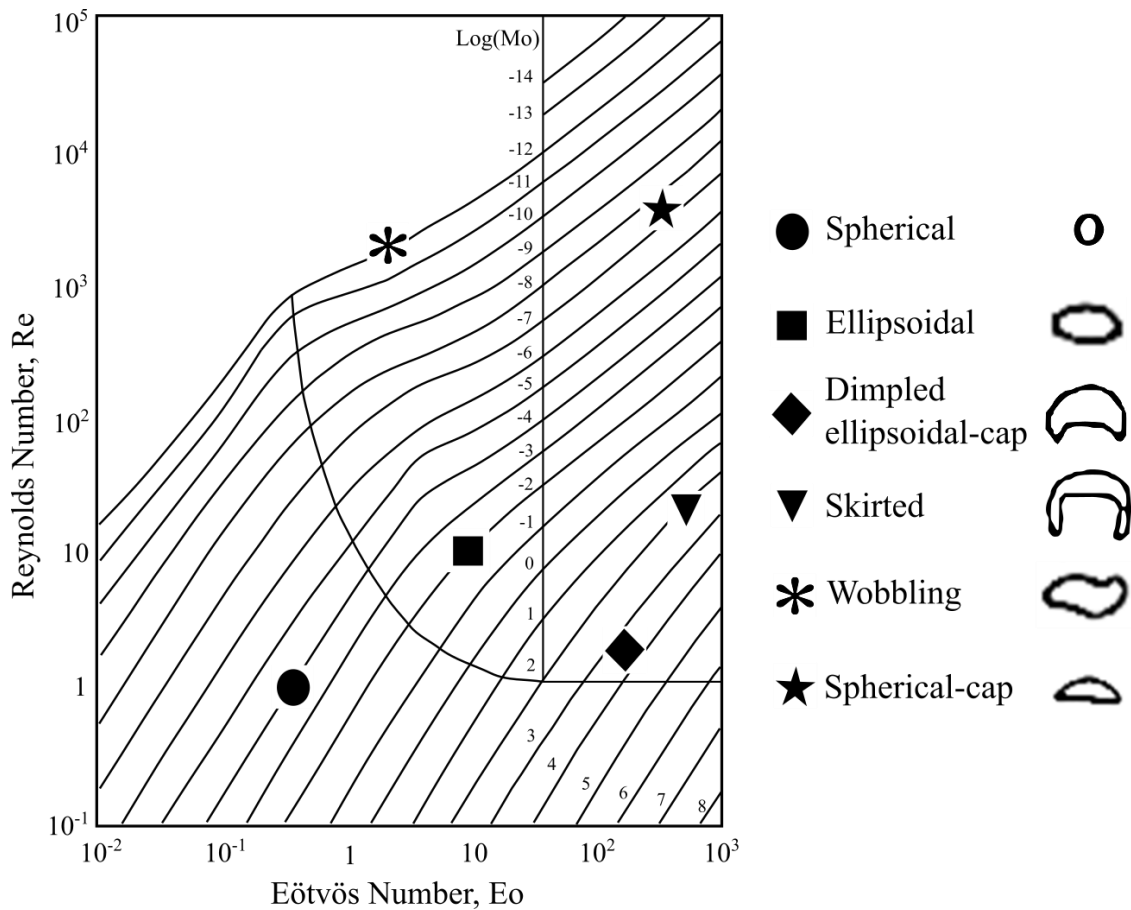
239 increases, resulting in greater interaction between the bubbles, with collisions, clusters formation,
240 and the occurrence of coalescence phenomena. The consequential appearance of larger bubbles
241 significantly alters the hydrodynamic scenario, with the concomitant presence of large (about
242 20 mm) and small bubbles. These latter rises rather fast ($1 - 2 \text{ m s}^{-1}$) stirring the liquid. The name of
243 churn flow (also known as heterogeneous flow) is due to the fact that the larger bubbles tend to
244 churn up the liquid [26]. In this state, as the corresponding Reynolds numbers prove to be higher
245 (see Table 3), spiraling and zigzagging motions can be observed. Moreover, due to this rather
246 turbulent environment, large bubbles often do not count on a clear definition of their form which
247 rather fluctuates quite casually. Nevertheless, some characteristic shapes can be identified, since the
248 morphology of the bubbles is mostly a function of the diameter, speed, and properties of the system.
249 The work carried out by Grace [27] produced a well-known generalized graphical correlation
250 (Figure 5) that depicts the individual geometry of a single rising bubble in terms of three
251 dimensionless numbers: Reynolds number, Eötvös number Eo and Morton number Mo defined as

$$Eo = \frac{g(\rho_l - \rho_g)d_b^2}{\sigma} \quad (9)$$

$$Mo = \frac{g\mu_l^4(\rho_l - \rho_g)}{\rho_l^2\sigma^3} \quad (10)$$

252 Note that, in this context, the characteristic length required to compute Reynolds number is d_b .

253



254

255 Figure 5. Grace's diagram [25]

256 By analyzing the diagram of Figure 5, oftentimes called Grace's diagram, one can surmise that the
 257 preferable shape is that of a spherical cap, given the high ratio between the exchange surface and
 258 the occupied volume. Proceeding with the reasoning, a zigzagging trajectory seems to be preferable,
 259 as it would extend the permanence time of the bubbles in the riser.

260 The experimental survey reported, in the chosen incipient churn regime, bubbles having an average
 261 mean diameter of 7 mm with shapes that varied from spherical, especially at the bottom of the riser
 262 (as it is to be expected since the orifices from which gas is introduced are circular), to spherical caps
 263 (Figure 6).

264 Table 3 summarizes the values of the main variables measured during the hydrodynamic
 265 experimental campaign and Table 4 reports the calculated variables and dimensionless numbers.

266

267

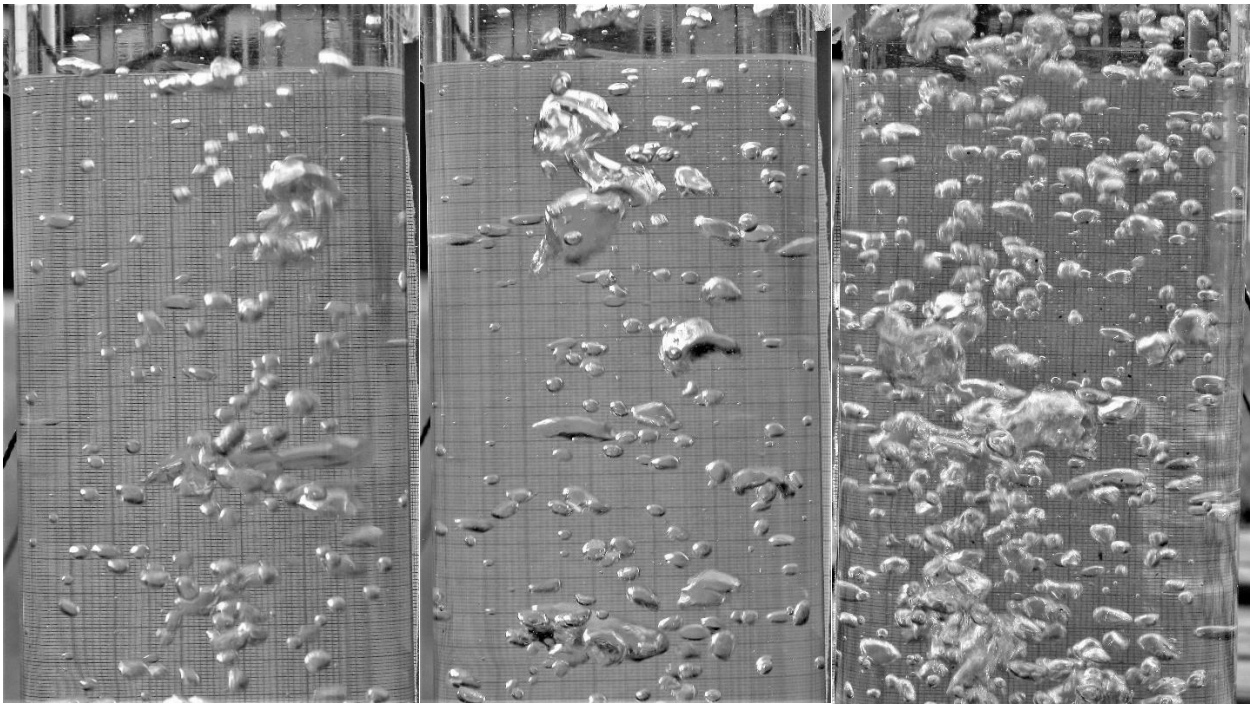
268 Table 3: Ranges of variables experimentally determined in ALRs.

Regime	Gas bubble velocity (m s ⁻¹)	Mean bubble diameter (mm)	Downcomer liquid velocity (m s ⁻¹)	Downcomer residence time (s)	Average time cycle (s)	Bubbles in the riser
Churn	0.573 – 0.642	5 – 7	0.168 – 0.235	3.5 – 4.8	20.4 – 27.6	1500-3000
Bubble	0.090 – 0.131	3 – 4	0.301–0.390	2 – 2.6	18 –24.7	3400-4000

269 Table 4: Ranges of variables calculated with empirical correlations and related dimensionless
270 numbers in ALRs.

Regime	Riser liquid velocity (m s ⁻¹)	Re _{riser}	Re _{down}	Bubbles per cross-section
Churn	0.084 – 0.117	7865 –10955	9438 – 13202	15 – 35
Bubble	0.056 – 0.057	5244 – 5337	14543 – 18843	35 – 50

271



272

273 a

b

c

274 Figure 6. Bubbles in the riser from the experimental campaign in incipient churn flow: (a) bubble
275 flow (gas bubble velocity = 0.12 m s⁻¹); (b) transition (gas bubble velocity = 0.35 m s⁻¹); (c) churn
276 flow (gas bubble velocity = 0.62 m s⁻¹)

277

278 **3.2 Evaluation of gas-liquid mass transport coefficients:**

279 Carbon dioxide contained in the flue-gas is the main substrate to microalgae needed for
280 photosynthesis, while oxygen is the main product. We focalized here on CO₂ since the proposed
281 multiphase model can describe in detail the behavior and the shape of the bubbles containing CO₂.
282 The main goal in controlling the shape of the bubbles and their residence time is to increase mass
283 transport between the bubbles and the liquid. Hence mass transport characteristics of the dissolved
284 CO₂ to the suspended microalgae in the liquid phase should be optimized in the operation of the
285 ALRs. The average mass flux of carbon dioxide to the microalgae is a function of the diffusion
286 coefficient D (m² s⁻¹) of CO₂ in water, the bulk concentration of CO₂ in water, the properties of the
287 fluid (viscosity and density), the dimension of microalgae and their velocity with respect to that of
288 the fluid.

289 The Sherwood number was calculated for the ALRs in the conditions reported in Table 5 as $Sh =$
290 $\frac{K_{oL}d_b}{D} = \frac{k_L a d_b^2}{D}$, where a (m² m⁻³) is the specific surface of the bubbles, d_b is the mean bubble
291 diameter and k_L is the liquid global mass transport coefficient.

292 $k_L a$ can be experimentally estimated in ALRs using different correlations proposed in the literature.

293 We adopted the following [28]:

$$k_L a = 0.79 \left(1 + \frac{A_{\text{down}}}{A_{\text{riser}}} \right)^{-2} u_{G,\text{riser}}^{0.8} \quad (11)$$

294
295 where A (m²) is the cross-sectional area and u_G (m s⁻¹) is the gas velocity.

296 The values of $k_L a$ for the incipient churn regimes experimentally tested are in the range 0.15 - 0.30.

297 The resulting Sh number ranges between 1200 and 3500. Since a can be also calculated as the product
298 of the number of bubbles in the riser and the mean surface of the bubble divided by the volume of the
299 riser, it results in a range of 7 - 15. The corresponding k_L is comprised between 0.01 and 0.04. The
300 values of k_L related to the bubble regime and the fully developed churn regime also tested during this

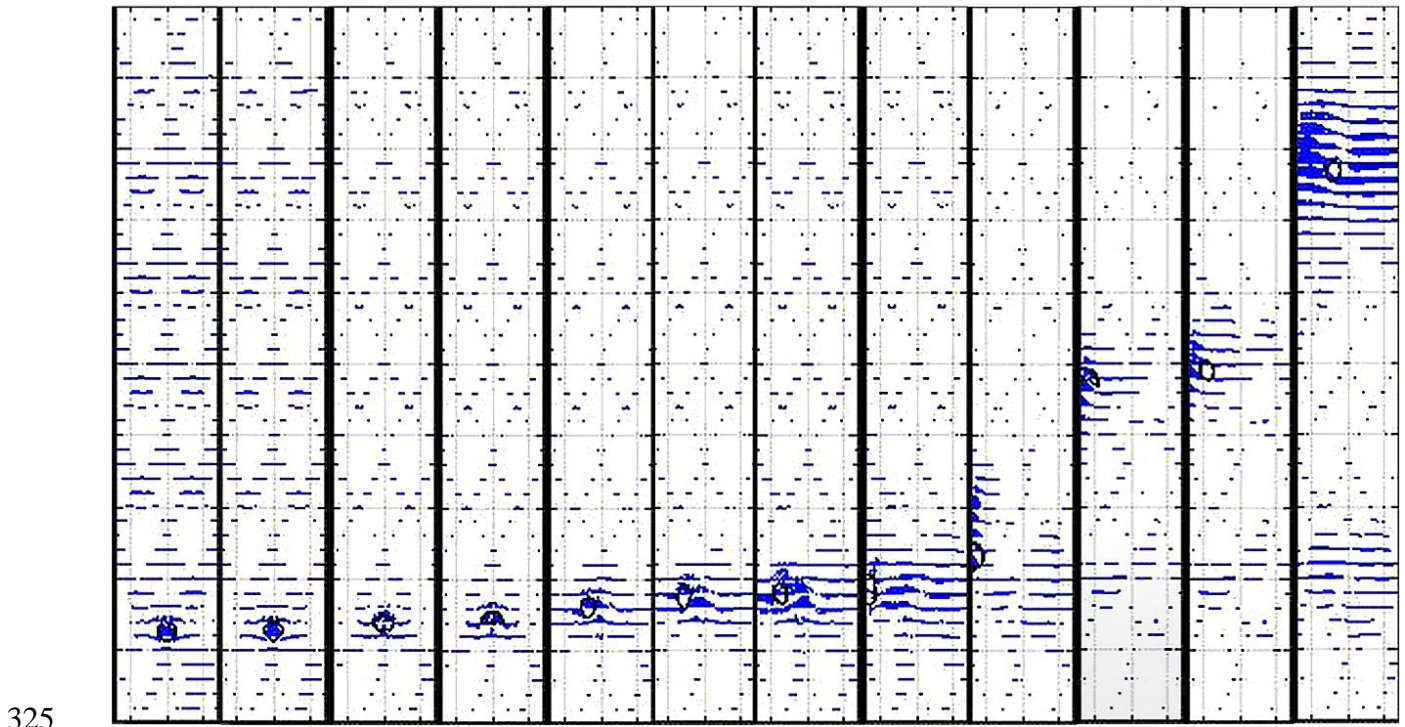
301 campaign, can be found in [13]. The optimal Sh number, calculated for the conditions reported in
302 Table 5, and chosen for the outdoor cultivation tests, is in the range 1400 - 1800.

303 **3.3. Simulation results using LSM**

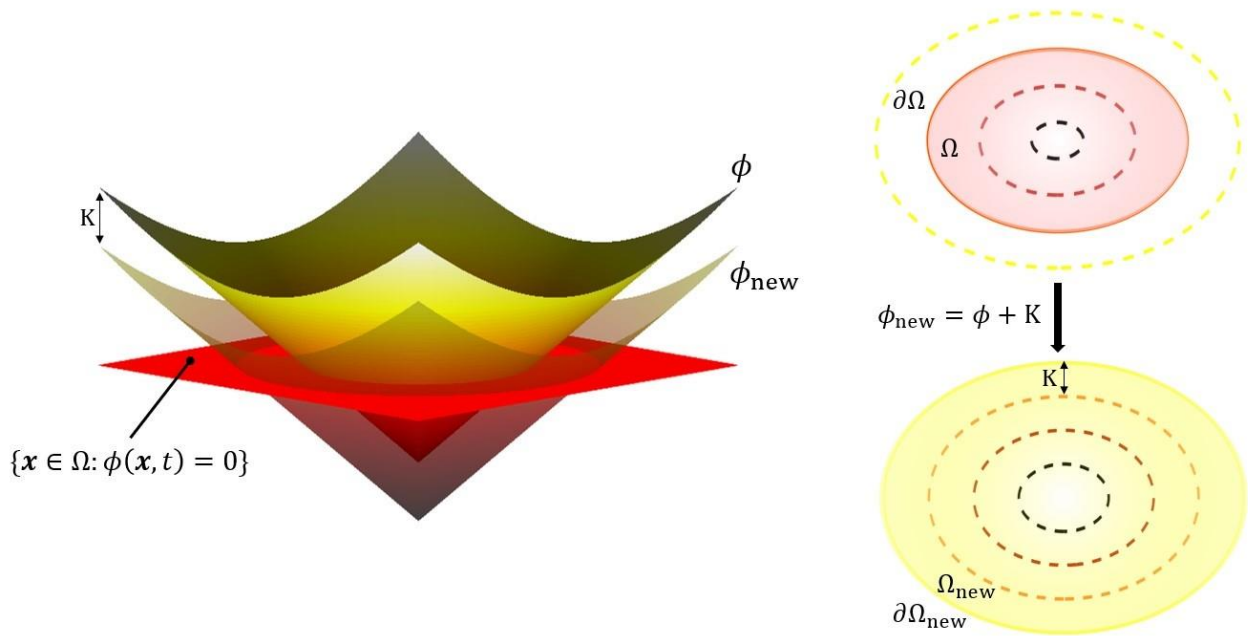
304 The overall computational model has been developed by gradually increasing the complexity of the
305 simulated system: first, we followed a single bubble, then we considered multiple puffs of bubbles
306 inserted in the computational domain at different times.

307 By virtue of the well-known volume loss problem by which the level set method is intrinsically
308 affected, it has been necessary to introduce a correction apt to remedy such an issue. With a mere
309 relocation of the level set function, lowered or raised according to need, using a simple algorithm
310 based on the bisection method, the simulation resulted more than satisfactory, as it can be noticed in
311 Figure 7. Substantially, one can exploit the fact that, provided that the scheme for the convection of
312 ϕ is sufficiently accurate, the error on the volume balance in each time step should be very small.
313 To avoid the error accumulation, which instead leads to the significant observed losses, it is thus
314 advisable to apply the correction at each time step, at the same time paying attention not to alter the
315 shape of the front. Presupposing a suitable rate of reinitialization of ϕ , in the proximity of the front,
316 the level set function is the signed distance and therefore, in the vicinity of its zero iso-contour
317 lines, ϕ presents level sets approximately equidistant to each other. Then, by translating ϕ upward
318 or downward by a signed constant K, which represents the distance between the original level set
319 and the one after the translation, the volume (or area in two dimensions) occupied by the gas is
320 conserved and the shapes of the interfaces are essentially unaltered (Figure 8). For this method to be
321 reliable, K must be small [29]. The algorithm counts the number of cells inside the interface at two
322 successive times to judge if the latter has increased or decreased and therefore if it is necessary to

323 lower or raise the zero-level set. The value of K is expected to be between zero and the maximum of
324 the zero-level set, so one possibility is to use the bisection method within these two extremes.



325
326 Figure 6. Simulation frames of the single bubble dynamic in the riser with LSM modified by the
327 relocation of the level set function using the bisection method. On the background, the speed plot of
328 the liquid velocity field; vectors are auto-scaled as to not overlap.

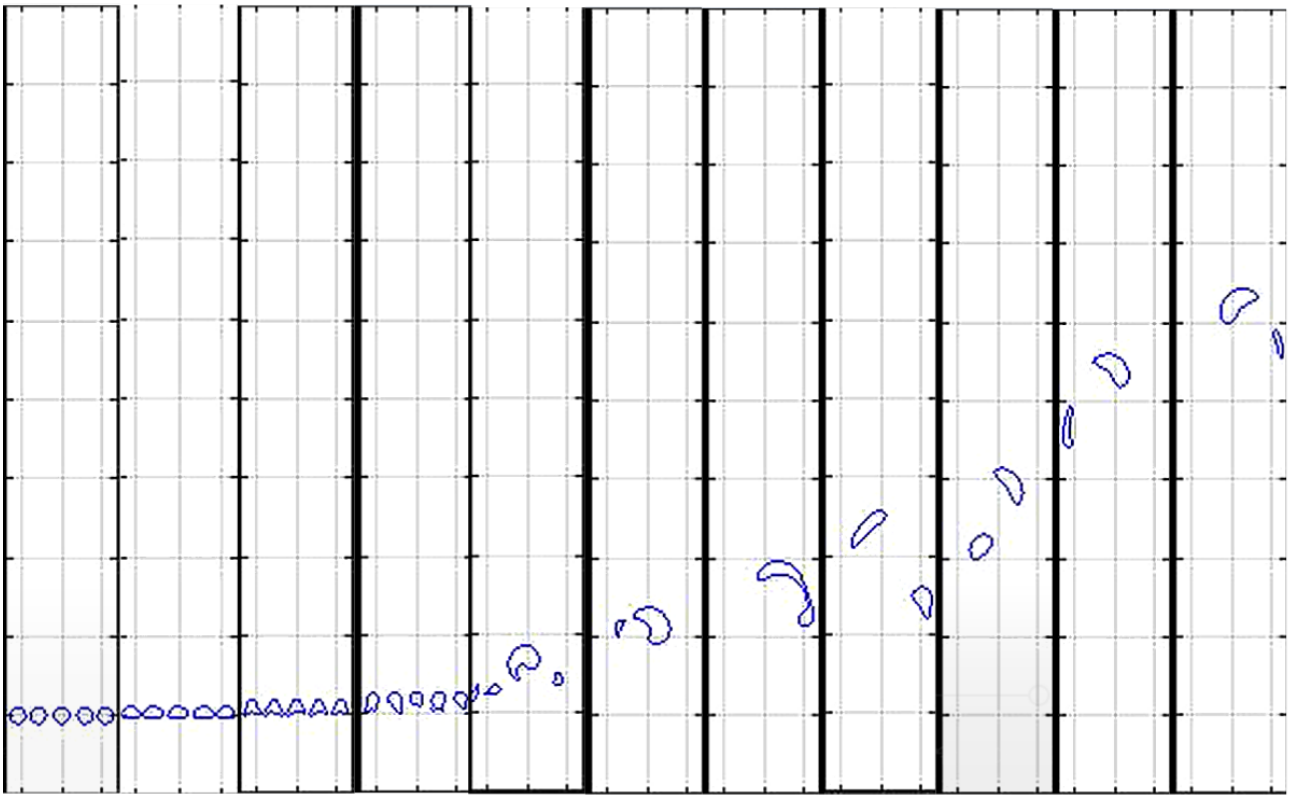


329

330 Figure 7. Level set correction by the relocation of the level set function ϕ , translated upward or
 331 downward by a signed constant K .

332 Although the precaution of inserting a correction on the balance of front volume worked just fine
 333 for the case of a single bubble, a further problem of distribution manifested in that of multiple
 334 bubbles: the number of grid cells standing within the fronts is preserved but the lost cells are
 335 typically added to the larger bubbles, without taking into account their position. The effect that
 336 follows is a sort of instantaneous transfer of matter, obviously devoid of physical meaning. It is
 337 possible to circumvent this issue (Figure 9) as long as one simulates the evolution of multiple
 338 bubbles that begin their walk at the same time and the frequency with which the correction is
 339 performed is suitably reduced [30]. Nevertheless, since this route is not feasible, it is necessary to

340 implement a different approach. With this in mind, the adoption of the so-called conservative level
341 set method (CLSM) [31,32] was decided.

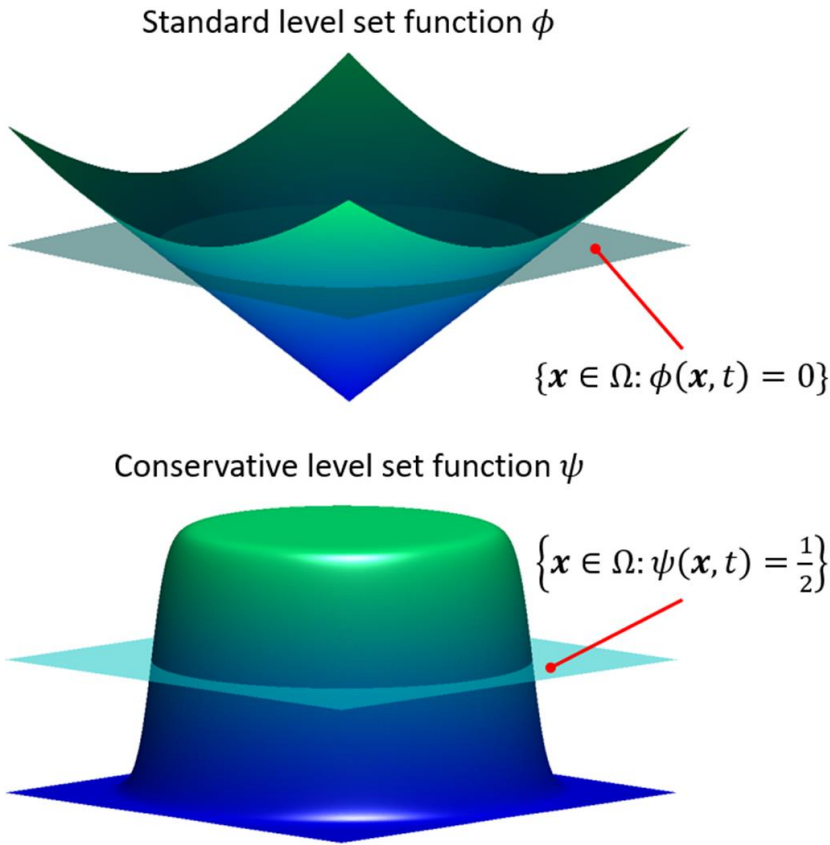


342
343 Figure 8. Simulation frames of instantaneous puff behavior for a 5 bubbles flow inside the riser,
344 with LSM modified by the relocation of the level set function using the bisection method. To obtain
345 a clearer and more understandable picture of bubble dynamics, the quiver plot has been disabled.

346 **3.4. Simulation results using CLSM**

347 CLSM entails the adoption of a different phase field function, employing ψ instead of ϕ , being
348 $\psi(\mathbf{x}, t) = H_\varepsilon(\phi(\mathbf{x}, t))$. In this way, the position of the interface is implicitly represented, avoiding
349 the need for computing both the step function and δ_ε . More importantly, if the advection of ψ is
350 carried out in a conservative way, the volume bounded by the $\frac{1}{2}$ iso-surface is approximately

351 preserved (exactly if one uses H instead of H_ε , but numerical issues would arise due to the nature of
 352 the sharp interface).



353

354 Figure 9. Comparison between standard LS (top) and Conservative LS (bottom) functions.

355 The function ψ is initialized as a hyperbolic tangent (Eq. (12)) of thickness ε as, for $\varepsilon \rightarrow 0$, it tends
 356 [33] to the exact Heaviside step function $H(\phi)$:

$$\psi(\mathbf{x}, t) = \frac{1}{1 + \exp\left(\frac{\phi(\mathbf{x}, t)}{\varepsilon}\right)} = \frac{1}{2} \left(1 + \tanh\left(\frac{\phi(\mathbf{x}, t)}{2\varepsilon}\right) \right) \quad (12)$$

357 The conservative level set method articulates in two steps. The first one, similarly to standard LSM,
 358 is the advection of the phase function, expressed as:

$$\frac{\partial \psi}{\partial t} + \mathbf{u} \cdot \nabla \psi = 0 \quad (13)$$

359 which, in the presence of a divergence-free flow field, one can reformulate as:

$$\frac{\partial \psi}{\partial t} + \nabla \cdot (\mathbf{u}\psi) = 0 \quad (14)$$

360 Eq. (14) is a conservation equation and as such lends conservation properties to the algorithm.

361 The second step is the reinitialization, needed to regularize the shape of ψ and enhance numerical

362 robustness. It consists of the resolution of the hyperbolic PDE Eq. (15), comprised by a

363 compressive limiter intended to sharpen the profile and a diffusive one in the normal direction, used

364 to balance it and maintain the adequate interface thickness.

$$\frac{\partial \psi}{\partial \tau} + \nabla \cdot (\psi(1 - \psi)\hat{\mathbf{n}}) = \varepsilon \nabla \cdot ((\nabla \psi \cdot \hat{\mathbf{n}})\hat{\mathbf{n}}) \quad (15)$$

365 The evolution equation Eq. (15), reported in conservative form, has to be solved to steady-state in

366 an artificial time τ framework. Courant–Friedrichs–Lewy (CFL) condition ensues for both

367 advection and re-initialization steps. In particular, when applying the artificial compression

368 technique, Olsson and Kreiss [31] suggested that

$$\Delta \tau \leq \frac{C(\Delta x)^2}{\varepsilon} = 2C(\Delta x)^{1+d} \quad (16)$$

369 with Courant number $C = 0.25$, ε interface thickness and d either 0 or very small (e.g. 0.1) for

370 complicated flows. There, mesh size Δx was assumed uniform. In this work, we chose to use

371 $\min(\Delta x, \Delta y)$ instead of Δx when dealing with the CFL condition in the rectangular space domain

372 whilst exactly Δx when Ω is a square. When implementing the CLSM, we adopted precisely this

373 latter definition of Ω , for the reasons already explained in section § 2.3.

374 Besides, concerning the choice of the mesh size, a tradeoff between volume conservation and

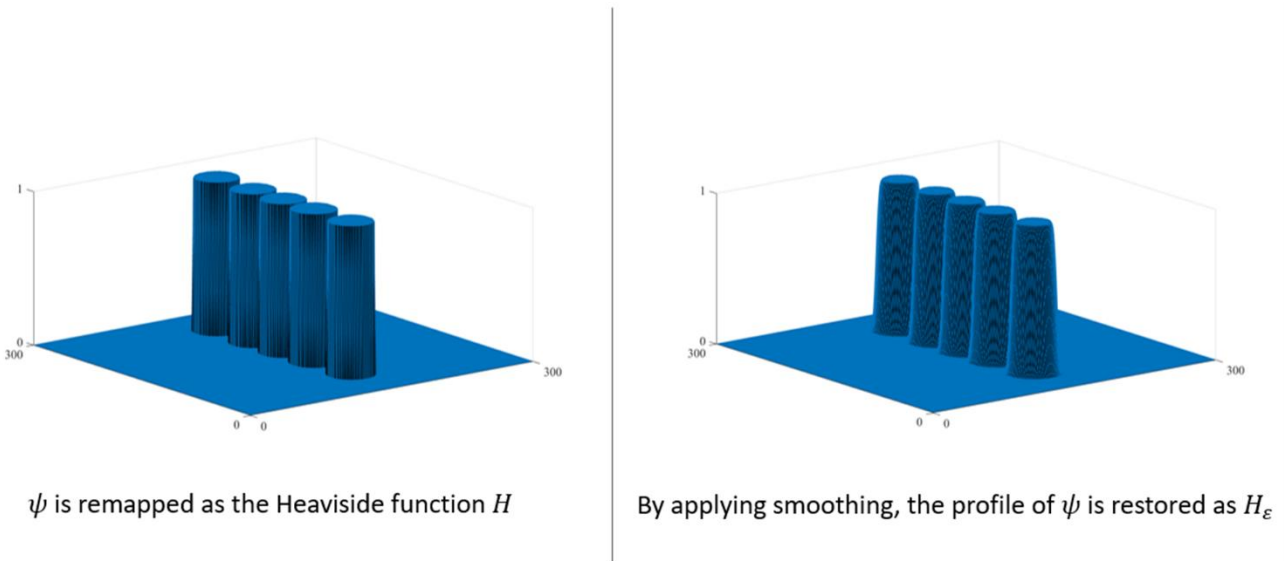
375 accuracy emerges: if on the one hand smaller ε (Eq. (17)) implies better volume preservation and

376 compliance with the CFL condition, on the other, it entails a decrease in the order of accuracy.

377 Exceeding in setting ε to a small value can determine the formation of spurious oscillations, which
 378 in turn may damage the normal field by switching its direction.

$$\varepsilon = \frac{(\Delta x)^{1-d}}{2} \quad (17)$$

379 In their presentation of CLSM, Olsson and Kreiss [31] attacked this problem by resorting to a
 380 second-order TVD method with Superbee limiter; nevertheless, in this way, the overall accuracy of
 381 the method is affected [34]. As efficaciously explained by McCaslin & Desjardins [35], excessive
 382 re-initialization can damage the simulation results revealing equation stiffness. To relieve this
 383 degradation, Desjardins et al. [36] suggested a reconstruction of ϕ from ψ through the fast
 384 marching method (FMM), by which a smooth normal field is obtained. Anyway, instead of
 385 resorting to this algorithm, called accurate conservative level set (ACLS), we privileged simplicity
 386 over computational speed: we opted for a remapping of ψ by first restoring the steep profile of the
 387 Heaviside step function, and then smearing it out with a Gaussian filter (Figure 11) [37]. Since this
 388 causes volume loss, in order not to waste the benefits of using CLS, we coupled it with the
 389 correction by translation of the level set function already used in the LSM.

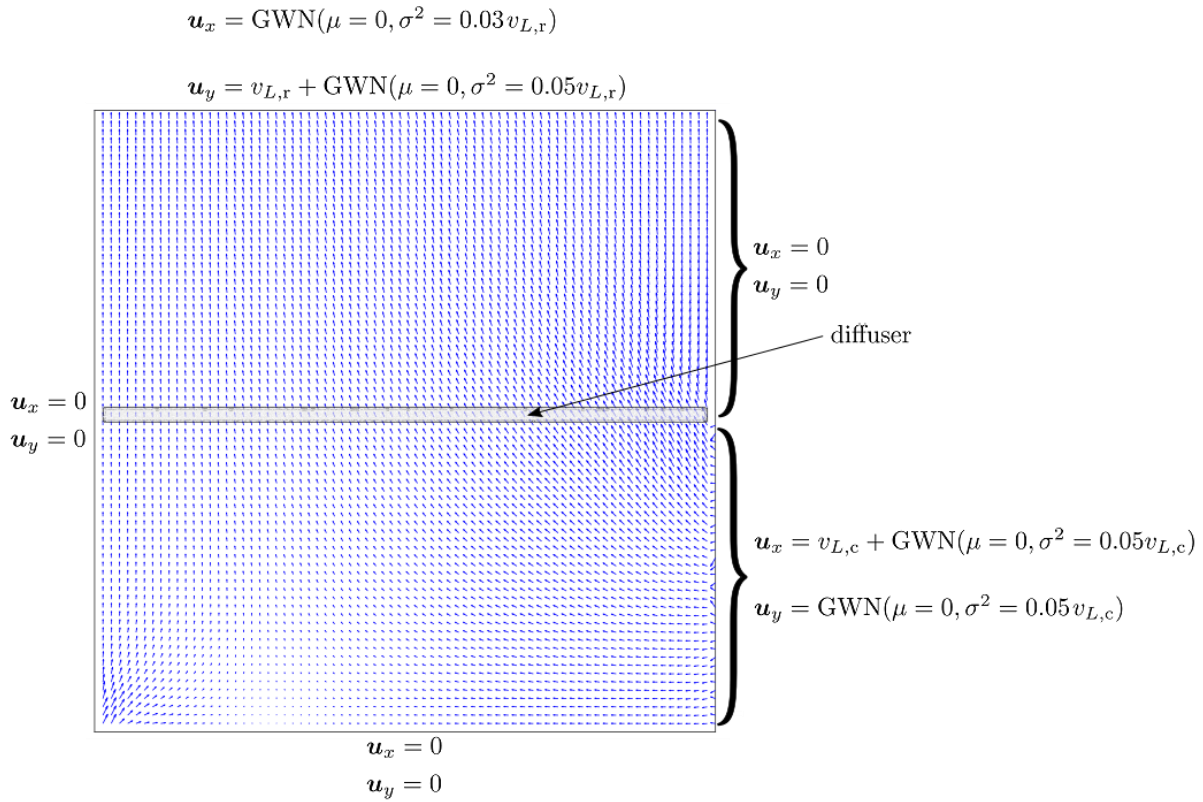


390

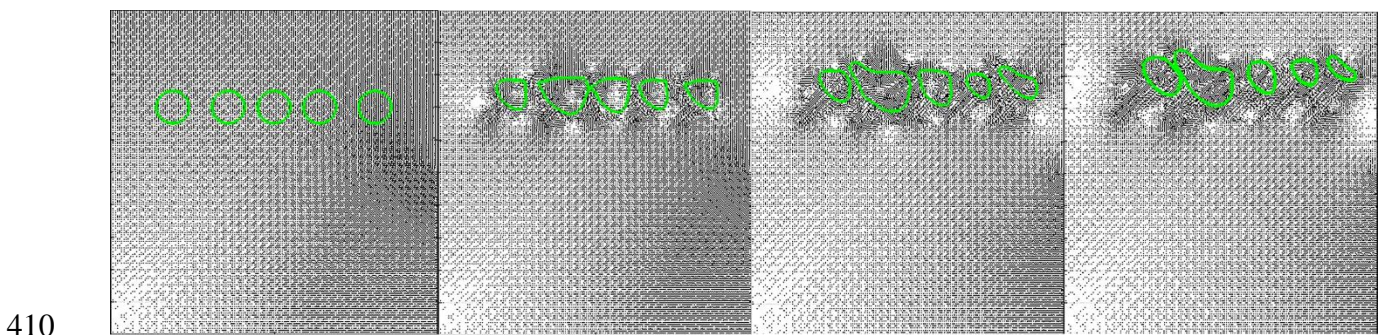
391 Figure 10: CLS function remapping.

392 Simulations were carried out on a regular square grid mad up by 300×300 points, as $\Omega = d_r \times d_r$,
393 with Dirichlet boundary conditions for each edge of the computational domain. In particular, no-slip
394 conditions were applied in both directions to the southern and western edges, to the upper half on
395 the eastern boundary. The definition is instead trickier on the northern boundary. We opted to
396 impose vertical component velocity of the same value as the mean liquid velocity measured
397 experimentally in the riser $v_{L,r}$ added with Gaussian White Noise (GWN, zero mean, $0.05 \cdot v_{L,r}$
398 variance), whilst pure GWN was used to introduce small aleatory deviations in the x -direction. The
399 same stratagem was employed to model the components of the imposed velocity profile on the
400 eastern edge in the lower half (fluid coming from the horizontal collector). In particular, \mathbf{u}_y is
401 described as GWN and \mathbf{u}_x the sum of mean liquid velocity in the collector $v_{L,c}$ and GWN. The
402 resulting velocity field, for one of the simulated diffuser positions, is depicted in Figure 12. The
403 detail of some simulations with CSLM, representing the bottom section of the riser, are presented in
404 Figure 13. Frames are reported with a frequency equal to 25Hz in order to better observe bubbles

405 shape dynamics (the mean experimental gas velocity into the riser is in the range $0.4 - 0.8 \text{ m s}^{-1}$ in
 406 incipient churn regime, see Table 3).



408 Figure 11: Velocity field at simulation starting time. The position of the diffuser assumed for this
 409 simulation is highlighted. For the sake of clarity, the grid is 150×150 instead of 300×300 .



411 Figure 12. Simulation of instantaneous puff behavior for a 5 bubbles flow inside the riser, with
 412 CLSM.

413 3.5. Validation of the model and optimal growing conditions

414 The hydrodynamic experimental tests were carried out to investigate the three different regimes
415 (bubble flow, transition or incipient churn, and churn) in the ALRs, in order to better characterize
416 the incipient churn flow, i.e. the regime suggested in the literature for microalgae nurturing in ALRs
417 [38,39] since it permits high Sh numbers without damaging the microalgae. In this regime the
418 concomitant presence of large and small bubbles was experimentally observed, where the larger
419 ones tend to slightly churn up the liquid while the smaller ones rise faster, stirring the liquid. The
420 results of simulations using the modified level set method confirmed the trajectories of the churning
421 bubbles at the corresponding conditions. Moreover, even if some adjustments need yet to be tested
422 with the proposed CLSM, the modeled shape of the bubbles strongly corresponds (see the spherical
423 caps in Figure 6b and Figure 9). So, the proposed model was used for the fine-tuning of the best-
424 operating conditions in the ALRs. In particular, many simulations in the transition regime between
425 bubble and churn were carried out. Different couples of Re numbers (respectively in the riser and in
426 the downcomer) lead the ALRs into this transition regime, with the difference among them being
427 the shape of the bubbles, which depends on the Eo number.

428 We choose an incipient churn regime with as similar as possible Re numbers in the two main
429 sections of the reactor and with the requested shape for the bubbles, i.e. small spherical bubbles
430 together with mainly spherical caps. These optimal conditions were kept in the outdoor cultivation
431 tests (Table 5). The microalgae related parameters measured inside the ALRs, outdoor operated
432 under these optimal hydrodynamic conditions, are also reported in Table 5 (and fully described in
433 [2]).

434 Table: Growing conditions in the outdoor cultivation tests

Cultivation parameters	Value	Hydrodynamics and mass transport parameters	Value
CO ₂ (%)	10	Re [downcomer]	11230
Glycerol (g L ⁻¹)	10	Re [riser]	10300

T ($^{\circ}\text{C}$)	12 - 25	Sh [CO_2]	1400 - 1800
pH	7 - 9.5	liquid velocity [downcomer] (m s^{-1})	0.202
Conductivity (μS)	700 - 1200	liquid velocity [riser] (m s^{-1})	0.111
DO_x (mg L^{-1})	10 - 19	Gas velocity (bubbles) (m s^{-1})	0.585
I_0 ($\mu\text{mol m}^2 \text{s}^{-1}$)	1000 - 1680	Mean bubble diameter (m)	0.005
PFD ($\mu\text{mol m}^2 \text{s}^{-1}$)	120 - 170	Bubbles in the riser	2200*
c_0 (g L^{-1}) initial, at each cycle	0.5	Average residence time (s)	26
c_f (g L^{-1}) at extraction	1.45 - 1.52	Gas flowrate ($\text{m}^3 \text{s}^{-1}$)	0.0015

435 *the estimated experimental error is 22%

436

437 **4. Conclusions**

438 This paper dealt with the simulation of the bubble dynamics within the riser of an external loop
439 airlift reactor, taking into account the geometry, and in particular the position of the sparger and the
440 horizontal collectors. All the data used for validating simulation were collected or calculated from a
441 set of experimental hydrodynamic campaigns on pilot-scale reactors. While classic CFD-based
442 Euler-Euler and, sometimes, Euler-Lagrange methods have been extensively employed to model
443 various types of reactors including bubble column ones, very few studies have been performed with
444 front-tracking techniques, especially in conjunction with multiple, small bubbles. Therefore, we do
445 not have many benchmarks for an effective comparison. These first results confirm the good ability
446 of the level set method to manage the complex dynamics of multiphase systems. In particular,
447 although simulating more bubbles at the same time, we identified bubble shapes coherent with those
448 belonging to the classes found experimentally (spherical, ellipsoidal, dimpled ellipsoidal-cap,
449 skirted, spherical-cap) in bubble, incipient churn, and churn regimes. Besides, the correct definition
450 of the characteristics of the flow field highlighted the zig-zagging trajectories desired and obtained
451 in the experimental tests. The physically-based model, after validation, was used for the fine-tuning
452 of the regime inside the ALRs. The final suggested operating conditions for the outdoor cultivation

453 tests entail Re numbers around 10000 - 11000, similar in all the sections of the ALRs; an Sh
454 number in the range 1400 - 1800 and spherical-caps in the upper half of the riser, slightly churning
455 the microalgae without damaging them. These conditions are reached by insufflating a flue gas
456 flowrate of $0.0015 \text{ m}^3 \text{ s}^{-1}$ in each ALR.

457 Anticipated future developments concern, on the one hand, further optimization of the management
458 of the computational domain and the reinitialization, and on the other hand the implementation of
459 mass balance equations apt to simulate the absorption of CO_2 in the liquid, central for the growth of
460 microalgae.

461 **Funding:** This research did not receive any specific grant from funding agencies in the public,
462 commercial, or not-for-profit sectors.

463 **Author contributions:** MN, PB, OP Conceptualization; MN, PB Formal analysis; MN, PB, OP
464 Investigation; MN, PB, OP Methodology; OP, PB Resources; MN Software; PB, OP Supervision;
465 PB, OP Validation; MN, OP Writing - original draft; MN, PB, OP Writing - review & editing.

466 **Notation**

a	specific surface of the bubbles	$\text{m}^2 \text{ m}^{-3}$
A	cross-sectional area	m^2
d	signed distance function	m
d_b	equivalent diameter of a bubble	m
d_c	diameter of the horizontal collectors of the ALR.	m
d_r	diameter of the riser of the ALR.	m
D	diffusion coefficient	$\text{m}^2 \text{ s}^{-1}$

$\hat{\mathbf{e}}_g$	gravitational acceleration unit vector	-
g	gravitational acceleration	m s^{-2}
\mathbf{g}	gravity	m s^{-2}
H	Heaviside step function	-
H_ε	smoothed Heaviside function	-
k_L	liquid global mass transport coefficient	m s^{-1}
L	characteristic length	m
L_r	length of the riser	m
$\hat{\mathbf{n}}$	unit normal vector to the interface	-
p	pressure	Pa
t	time	s
T	convective time	s
u_G	gas velocity	m s^{-1}
\mathbf{u}	velocity	m s^{-1}
U	characteristic speed	m s^{-1}
$v_{L,c}$	liquid mean velocity in the horizontal collector	m s^{-1}
$v_{L,r}$	liquid mean velocity in the riser	m s^{-1}

\mathbf{x}	position vector	m
\mathcal{S}	modified signed function	m

467 *Greek letters*

$\delta(\phi)$	Dirac delta distribution	m^{-1}
$\delta_\varepsilon(\phi)$	mollified delta function	m^{-1}
ε	numerical smearing coefficient; interface thickness	-
ζ	volume viscosity	Pa s
κ	curvature	m^{-1}
K	level set method correction parameter	m
μ	dynamic viscosity	Pa s
ν	kinematic viscosity	$\text{m}^2 \text{s}^{-1}$
ξ	body force per unit volume	N m^{-3}
ϕ	standard level set function	m
ρ	mass density	kg m^{-3}
σ	surface tension coefficient	N m^{-1}
τ	artificial time	s
ψ	conservative level set function	m

Ω	computational domain	-
$\partial\Omega$	boundary between Ω^+ and Ω^-	-

468 *Dimensionless numbers*

Eo	Eötvös number
Fr	Froude number
Mo	Morton number
Re	Reynolds number
Sh	Sherwood number

469 *Subscripts and superscripts*

*	dimensionless
<i>l</i>	liquid phase
<i>g</i>	gas phase

470 *Operators*

$\frac{\partial}{\partial t}$	partial time derivative	s^{-1}
$\frac{D}{Dt}$	substantial derivative	s^{-1}
∇	gradient operator	m^{-1}
$\nabla \cdot$	divergence operator	m^{-1}

Δ

Laplacian

 m^{-2}

471 **Supplementary Materials**

472 **S1. NSE non-dimensionalization**

473 In section 2.2, a non-dimensional version of the NSE is presented. In the following, the procedure to
474 obtain such a formulation is expounded.

475 To complete the non-dimensionalization of the NSE, as it is deducible from dimensional analysis, it
476 is necessary to multiply both sides of Eq. (1) by the constant term $L U^{-2}$, hence obtaining the non-
477 dimensionalized Navier–Stokes equations for an incompressible, isothermal, Newtonian fluid.

$$\frac{D\mathbf{u}^*}{Dt^*} = -\nabla^* p^* + \frac{gL}{U^2} \mathbf{g}^* + \frac{\nu}{LU} \Delta^* \mathbf{u}^* \quad (\text{S.1})$$

478 As it is well known, for Newtonian, incompressible fluids, the continuity equation applies. It
479 describes the velocity field \mathbf{u} as a solenoidal vector field:

$$\nabla \cdot \mathbf{u} = 0 \quad (\text{S.2})$$

480 It follows that, if one considers $\boldsymbol{\xi} = \rho \mathbf{g}$ as it is often the case,

$$\frac{D\mathbf{u}}{Dt} = -\frac{1}{\rho} \nabla p + \mathbf{g} + \frac{\mu}{\rho} \Delta \mathbf{u} = -\frac{1}{\rho} \nabla p + \mathbf{g} + \nu \Delta \mathbf{u} \quad (\text{S.3})$$

481 having introduced the kinematic viscosity $\nu = \mu \rho^{-1}$ as well.

482 To perform non-dimensionalization, some scaling parameters are needed. One way to attack the
483 problem [40] is to choose some characteristic length L and characteristic speed U and, exploiting
484 gravitational acceleration g , to define suitable non-dimensional variables as follows:

$$\mathbf{x}^* = \frac{\mathbf{x}}{L}, \quad \mathbf{u}^* = \frac{\mathbf{u}}{U}, \quad \mathbf{g}^* = \hat{\mathbf{e}}_g = \frac{\mathbf{g}}{g}$$

485 The selection of a proper pressure scale is not a foregone conclusion: a possibility, advisable for the
 486 cases in which dynamic effects are dominant, consists of obtaining the non-dimensional pressure p^*
 487 as the ratio $p \rho^{-1} U^{-2}$. Besides, it is easy to observe how $\nabla^* = L \nabla$ and that the introduction of L and
 488 U naturally leads to the identification of the time scale, since

$$T = \frac{L}{U} \quad (\text{S.4})$$

489 where T is the convective time. It follows that one can define t^* as $t^* = t T^{-1}$.

490 L and U are adopted to assemble dimensionless quantities useful to analyze the similarity of
 491 different systems. For the similarity constraint to be satisfied, geometrical similarity, as well as
 492 similarity of boundary conditions and equality of each of these dimensionless groups are required.

493 Generally speaking, characteristic physical quantities are liable to arbitrary choice, provided that
 494 they are well defined and referred to the same geometrical locations for every system [41].

495 Rearranging these latter mathematical expressions defining the non-dimensional quantities and
 496 plugging them in Eq. (S.3) leads to

$$\frac{U^2}{L} \frac{D\mathbf{u}^*}{Dt^*} = -\frac{\rho U^2}{L} \nabla^* p^* \frac{1}{\rho} + g \mathbf{g}^* + \nu \frac{U}{L^2} \Delta^* \mathbf{u}^* \quad (\text{S.5})$$

497 being Δ^* the non-dimensional Laplacian, i.e. the square of ∇^* .

498 To complete the non-dimensionalization of the NSE, as it is deducible from dimensional analysis, it
 499 is necessary to multiply both sides of Eq. (S.5) by the constant term $L U^{-2}$, hence obtaining the non-
 500 dimensionalized Navier–Stokes equations for an incompressible, isothermal, Newtonian fluid.

$$\frac{D\mathbf{u}^*}{Dt^*} = -\nabla^* p^* + \frac{gL}{U^2} \mathbf{g}^* + \frac{\nu}{LU} \Delta^* \mathbf{u}^* \quad (\text{S.6})$$

501 Two well-known similarity numbers, i.e. dimensionless groups, emerge in this way: the Reynolds
 502 number Re and Froude number Fr.

$$\text{Re} = \frac{LU}{\nu} \quad (\text{S.7})$$

$$\text{Fr} = \frac{U}{\sqrt{gL}} \quad (\text{S.8})$$

503 Eq. (S.6) can, therefore, be condensed in

$$\frac{D\mathbf{u}^*}{Dt^*} = \frac{\partial \mathbf{u}^*}{\partial t^*} + \mathbf{u}^* \cdot \nabla^* \mathbf{u}^* = -\nabla^* p^* + \frac{1}{\text{Fr}^2} \hat{\mathbf{e}}_g + \frac{1}{\text{Re}} \Delta^* \mathbf{u}^* \quad (\text{S.9})$$

504 **S2. LSM details**

505 LSM is a front capturing method, depicting a picture in which coordinates are fixed in space, i.e. an
 506 Eulerian method. From the perspective of the simulation of a system that involves bubble motion
 507 within a liquid domain and requires a high degree of precision considering the importance of their
 508 shape, Eulerian methods seem to be preferable. In fact, having an external stationary reference
 509 frame at one's disposal offers undeniable advantages in the delineation of bubble shape evolution: it
 510 enables one to take into account topology variations whereby bubbles may coalesce and break up.
 511 Moreover, Lagrangian methods prove to be ill-suited to cope with such problems since, while they
 512 preserve a sharp interface representation, they demand re-meshing when large deformations
 513 manifest and are subjected to mesh tangling and numerical inaccuracy due to highly irregular
 514 meshes [42], being prone to blow up.

515 Among other virtues of LSM, its ability to naturally determine intrinsic geometrical properties of
 516 the moving interfaces such as their normal vector $\hat{\mathbf{n}}$ turns out to be very advantageous also in the
 517 perspective of dealing with physical properties such as surface tension. Simply by differentiation of
 518 ϕ , one is able to infer both $\hat{\mathbf{n}}$ and the curvature κ :

$$\hat{\mathbf{n}} = \frac{\nabla\phi}{\|\nabla\phi\|} \quad (\text{S.10})$$

$$\kappa = \nabla \cdot \hat{\mathbf{n}} \quad (\text{S.11})$$

519 *S2.1. On LSM reinitialization*

520 ϕ is generally initialized into the signed distance function $d(\mathbf{x}, t) = \min_{\mathbf{x}_{\partial\Omega} \in \partial\Omega} |\mathbf{x} - \mathbf{x}_{\partial\Omega}|$, with $\mathbf{x}_{\partial\Omega}$
521 being the closest point of the front from \mathbf{x} : in the probable event that the speed function \mathbf{u} is not
522 constant, ϕ can become either very flat or steep. Sussman et al. [43] proposed an iterative
523 reinitialization of ϕ by reformulating $d(\mathbf{x}, t)$, i.e. the unique viscosity solution of the Eikonal
524 equation $|\nabla\phi| = 1$ anchored at ϕ_0 , by solving another Hamilton-Jacobi PDE (Eq. (S.10)) in an
525 artificial time reference τ .

$$\frac{\partial\phi(\mathbf{x}, t)}{\partial\tau} + \mathcal{S}(|\nabla\phi(\mathbf{x}, t)| - 1) = 0 \quad (\text{S.12})$$

526 where \mathcal{S} is a modified signed function [44]. Albeit subject to CFL limitations, this procedure is
527 widely employed as it translates into an accurate reconstruction of the distance profile.

528 The application of LSM to fluid dynamic problems hinges on three mathematical functions: other
529 than the level set function ϕ , Dirac delta δ and Heaviside step function H play a crucial role. Even
530 though the thickness of the interface may be inconsiderable from a physical perspective, LSM calls
531 for the prescription of a fixed and numerically relevant front thickness as a means to alleviate
532 numerical difficulties that may arise due to sharp changes of the considered physical properties
533 across $\partial\Omega$. To avoid jump discontinuities, sharp changes in the properties of Ω^+ and Ω^- regions are
534 smudged by the means of a smoothed Heaviside function $H_\varepsilon(\phi)$, whilst mollified delta function
535 $\delta_\varepsilon(\phi)$ is employed to analogously model the surface tension force [45]. Specifically, Osher and
536 Fedkiw [46] report the definition of $H_\varepsilon(\phi)$ as

$$H_{\varepsilon}(\phi) = \begin{cases} 0 & \phi < -\varepsilon \\ \frac{1}{2} + \frac{\phi}{2\varepsilon} + \frac{1}{2} \sin\left(\frac{\pi\phi}{\varepsilon}\right) & -\varepsilon \leq \phi \leq \varepsilon \\ 1 & \varepsilon < \phi \end{cases} \quad (\text{S.13})$$

537 where ε is a parameter influencing the numerical smearing and usually of the same order of
 538 magnitude as the interface thickness, and $\delta_{\varepsilon}(\phi)$ as its derivative:

$$\delta_{\varepsilon}(\phi) = \begin{cases} 0 & \phi < -\varepsilon \\ \frac{1}{2\varepsilon} + \frac{1}{2\varepsilon} \cos\left(\frac{\pi\phi}{\varepsilon}\right) & -\varepsilon \leq \phi \leq \varepsilon \\ 1 & \varepsilon < \phi \end{cases} \quad (\text{S.14})$$

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CAPTIONS

662 **Figures**

663 Figure 1. External loop airlift reactor: A – riser; B – downcomer; C – horizontal collectors.

664 Figure 2. Illustration of the LSM.

665 Figure 3. Numerical model schematization.

666 Figure 4. Staggered grid with boundary cells.

667 Figure 5. Grace's diagram [40]

668 Figure 6. Bubbles in the riser from the experimental campaign in incipient churn flow: (a) bubble
669 flow (gas bubble velocity = 0.12 m s^{-1}); (b) transition (gas bubble velocity= 0.35 m s^{-1}); (c) churn
670 flow (gas bubble velocity = 0.62 m s^{-1})

671 Figure 13. Simulation frames of the single bubble dynamic in the riser with LSM modified by the
672 relocation of the level set function using the bisection method. On the background, the speed plot of
673 the liquid velocity field; vectors are auto-scaled as to not overlap.

674 Figure 14. Level set correction by the relocation of the level set function ϕ , translated upward or
675 downward by a signed constant K.

676 Figure 15. Simulation frames of instantaneous puff behavior for a 5 bubbles flow inside the riser,
677 with LSM modified by the relocation of the level set function using the bisection method. To obtain
678 a clearer and more understandable picture of bubble dynamics, the quiver plot has been disabled.

679 Figure 10. Comparison between standard LS (top) and Conservative LS (bottom) functions.

680 Figure 11: CLS function remapping.

681 Figure 12: Velocity field at simulation starting time. The position of the diffuser assumed for this
682 simulation is highlighted. For the sake of clarity, the grid is 150×150 instead of 300×300 .

683 Figure 13. Simulation of instantaneous puff behavior for a 5 bubbles flow inside the riser, with CLSM.

684 **Tables**

685 Table 1. Physical-chemical data used in the simulation.

686 Table 2. Data gathered in the experimental campaigns.

687 Table 3: Ranges of variables experimentally determined in ALRs.

688 Table 4: Ranges of variables calculated with empirical correlations and related dimensionless
689 numbers in ALRs.

690 Table 5: Growing conditions in the outdoor cultivation tests

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